

GENERAL NOTES

SYMBOLS AND ABBREVIATIONS:	UNITS
AC	ALTERNATING CURRENT
AGM	ACTUAL CUBIC FEET PER MINUTE
AH	AMPERE HOURS
BAR	BAR
BAR	BAR GAGE
BAR	BAR DIFFERENTIAL
BTU	BRITISH THERMAL UNIT
C	DEGREES CELSIUS
CFM	CUBIC FEET PER MINUTE
CS	CENTISTOKES (MM SQUARED PER SECOND)
DC	DIRECT CURRENT
DFT	DIFFERENTIAL
F	DEGREE FAHRENHEIT
FT	FEET
GAL	GALLONS
GPM	GALLONS PER MINUTE
HR	HOUR
KJ	KILOJOURN
KJ	KILOJOULE
KN	KILONEWTN
KPA	KILOPASCAL

NOTE: FLOW, PRESSURE AND TEMPERATURE VALUES SHOWN ON THIS DRAWING ARE INTENDED FOR USE IN THE DESIGN OF EXTERNAL SYSTEMS AND EQUIPMENT ONLY AND CANNOT BE USED IN ANY TURBINE PERFORMANCE CALCULATIONS.

1.1 SPECIFICATIONS: THE FOLLOWING SOLAR DOCUMENTS APPLY TO THIS DRAWING.

ES 9-64	FASTENER INSTALLATION AND TORQUE VALUES
ES 9-62	INGESTIVE CLEANING SOLAR GAS TURBINE ENGINES
ES 9-68	FUEL, AIR, AND WATER (OR STEAM) FOR SOLAR GAS TURBINE ENGINES
ES 9-224	LUBRICATING OIL FOR SOLAR GAS TURBINES
ES 9-414	ELEVATING AND LEVELING PACKAGE BASES
ES 2184	CLEANING AND FLUSHING OF HYDRAULIC SYSTEMS/COMPONENTS
ES 2201	AUXILIARY SERVICE AIR

INSTALLATION INSTRUCTIONS FOR:

ES 1748	LUBE OIL TANK VENT WITH SEPARATOR AND FLAME ARRESTOR
ES 1832	EXHAUST SILENCER FOR SOLAR TURBINE ENGINES

1.2 SYSTEM DRAWINGS:

75681-149543	AIR ASSIST/ENGINE DRAIN
75681-149544	LUBE OIL SYSTEM SCHEMATIC
75681-149545	FUEL SYSTEM SCHEMATIC
75681-149546	DIRECT AC START SYSTEM
75681-149222	AUXILIARY ARRANGEMENT INSTALLATION
75681-149223	TURBOMACHINERY MECHANICAL INSTALLATION
75681-149310	ELECTRICAL SCHEMATIC
75681-149311	WIRING DIAGRAM
75681-149312	INTERFACE/INTERCONNECT DIAGRAM

TURBINE EXHAUST SYSTEM

2.0 GENERAL EXHAUST SYSTEM DESIGN GUIDELINES:

A. SUPPORT OF THE EXHAUST SYSTEM SHOULD PREVENT DAMAGE TO SOLAR SUPPLIED COMPONENTS DUE TO THERMAL EXPANSION. SEE PACKAGE INSTALLATION SECTION OF THE TURBOMACHINERY MECHANICAL INSTALLATION NOTES FOR APPLICABLE PACKAGE UNITS AND CRITERIA.

B. MOUNTING BRACKETS MAY BE WELDED TO THE SHELL OF THE DUCTING, THE TYPICAL METAL THICKNESS AND TYPE IS 10 GAUGE, (0.134" (3.4 MM) THICK NOMINAL), ASTM A36/A569 UNLESS OTHERWISE STATED IN THESE NOTES.

C. THE EXPANSION JOINT, WILL NOT SUPPORT THE WEIGHT OF DUCTING OR COMPONENTS. THE ALLOWABLE DISPLACEMENTS OF THE EXPANSION JOINT ARE: AXIAL = 1/2" (38 MM), LATERAL = 1/2" (38 MM), RADIAL = 0 DEGREES. THE EXPANSION JOINT HAS NO APPRECIABLE SPRING RATE.

2.1 WHEN ASSEMBLING OR DISASSEMBLING THE SYSTEM, USE THE LIFTING LUGS PROVIDED TO HANDLE EACH COMPONENT INDIVIDUALLY. EMPLOY INDUSTRY ACCEPTED LIFTING TECHNIQUES WHEN LIFTING LUGS ARE NOT PROVIDED. REFER TO COMPONENT WEIGHTS AND ASSUME A GEOMETRIC CENTER OF GRAVITY. DO NOT USE THE LIFTING LUGS FOR PERMANENT MOUNTING. (SEE NOTE 2.0, PARAGRAPH B).

2.2 THE DRAIN CONNECTIONS SHOWN ON THE ANGLARY COMPONENTS ARE DESIGNED TO REMOVE EXCESSIVE AMOUNTS OF RAIN OR MOISTURE FROM THE SYSTEM DURING EXTENDED SHUT DOWN PERIODS. CONNECTION TO THESE DRAINS IS OPTIONAL FOR THE CUSTOMER. RAIN AND MOISTURE IS ALSO REMOVED VIA THE EXHAUST COLLECTOR DRAIN ON THE PACKAGE.

2.3 ATTACHING HARDWARE KITS ARE INCLUDED FOR ALL FLANGES IN SOLAR SCOPE OR THE TURBINE EXHAUST INTERFACE FLANGE, (AS APPLICABLE). HARDWARE IS NOT SUPPLIED FOR FOUNDATION OR SUPPORT STEEL INTERFACES, UNLESS INCLUDED IN SOLAR SCOPE.

2.4 THE STATIC PRESSURE DROP ACROSS ALL ITEMS SHOWN ON THIS INSTALLATION DRAWING IS 2.5" H₂O (64 MM). IF ADDITIONAL DUCTING OR DEVICES DONE BY OTHERS IS USED, THE PRESSURE DROP OF THAT DUCTING OR DEVICE SHOULD BE SUBTRACTED FROM THE ALLOWABLE DROP ALLOCATED FOR IN THE TURBINE PERFORMANCE GUARANTEE.

2.5 INDIVIDUAL COMPONENTS OF THE TURBINE EXHAUST SYSTEM ARE DESIGNED TO WITHSTAND A WINDLOAD OF 120 MI/HR (183 KM/HR) OR EQUIVALENT SEISMIC OR LATERAL ACCELERATION. ASSEMBLIES OF INDIVIDUAL COMPONENTS CANNOT BE CONSIDERED TO BE STRUCTURALLY ADEQUATE TO THE SAME LOADS WITHOUT ADDITIONAL ENGINEERING.

2.6 BARE METAL EXHAUST SURFACES TYPICALLY OPERATE AROUND 900° (482°C). PERSONNEL PROTECTION IS RECOMMENDED AND CARE MUST BE TAKEN IN THE LOCATION OF HOT EXHAUST SURFACES. DO NOT LOCATE EXHAUST COMPONENTS ON OR NEAR FLAMMABLE SURFACES.

2.7 THE ALLOWABLE FLANGE LOADS ON THE EXHAUST SILENCER, ARE 2000 LBS (892 KN) AXIAL, 750 LBS (3.3 KN) SHEAR AND 5400 LB/IN (30.6 KN/M) MOMENT. CARE SHOULD BE TAKEN TO ENSURE THAT THE STACK DESIGN DOES NOT EXCEED THESE LOADS.

TURBINE AIR INLET SYSTEM

3.0 A MINIMUM CLEARANCE OR CLEAR AREA IS REQUIRED TO SERVICE THE FILTER ELEMENTS AND TO ENSURE PROPER AIRFLOW TO THE TURBINE. DO NOT BLOCK DOOR SWINGS OR ACCESS AREAS.

3.1 MOUNTING BRACKETS MAY BE WELDED TO THE SHELL OF THE DUCTING, THE TYPICAL METAL THICKNESS AND TYPE IS 10 GAUGE, (0.134" (3.4 MM) THICK NOMINAL), ASTM A36/A569 UNLESS OTHERWISE STATED IN THESE NOTES. DO NOT WELD ON FILTERS OR SILENCERS. WELDING ON THESE DEVICES MAY RESULT IN FIRE OR DAMAGE TO THE DEVICE.

3.2 WHEN ASSEMBLING OR DISASSEMBLING THE SYSTEM, USE THE LIFTING LUGS PROVIDED TO HANDLE EACH COMPONENT INDIVIDUALLY. EMPLOY INDUSTRY ACCEPTED LIFTING TECHNIQUES WHEN LIFTING LUGS ARE NOT PROVIDED. REFER TO COMPONENT WEIGHTS AND ASSUME A GEOMETRIC CENTER OF GRAVITY. DO NOT USE THE LIFTING LUGS FOR PERMANENT MOUNTING. (SEE NOTE 3.1).

3.3 NOTE NOT USED.

3.4 ATTACHING HARDWARE KITS ARE INCLUDED FOR ALL FLANGES IN SOLAR SUPPLIED COMPONENTS OR THE TURBINE INLET INTERFACE FLANGE, (AS APPLICABLE). HARDWARE IS NOT SUPPLIED FOR ANY OTHER UPSTREAM CONNECTIONS. HARDWARE IS NOT SUPPLIED FOR FOUNDATION OR SUPPORT STEEL INTERFACES, UNLESS INCLUDED IN SOLAR SCOPE.

3.5 THE TOTAL PRESSURE DROP ACROSS ALL ITEMS SHOWN ON THIS INSTALLATION DRAWING IS 2.25" H₂O (57 MM H₂O). IF ADDITIONAL DUCTING OR DEVICES DONE BY OTHERS IS USED, THE PRESSURE DROP OF THAT DUCTING SHOULD BE SUBTRACTED FROM THE ALLOWABLE PRESSURE DROP IN THE TURBINE PERFORMANCE GUARANTEE.

3.6 INDIVIDUAL COMPONENTS OF THE TURBINE INLET SYSTEM ARE DESIGNED TO WITHSTAND A WINDLOAD OF 120 MI/HR (183 KM/HR) OR EQUIVALENT SEISMIC OR LATERAL ACCELERATION. ASSEMBLIES OF INDIVIDUAL COMPONENTS CANNOT BE CONSIDERED TO BE STRUCTURALLY ADEQUATE TO THE SAME LOADS WITHOUT ADDITIONAL ENGINEERING.

3.7 DUCTING DESIGN, IF DONE BY OTHERS, SHOULD FOLLOW API 618/RP 11-POT AND/OR SMACHA STANDARDS AND SPECIFICATIONS. RECOMMENDED WALL THICKNESS IS NOT LESS THAN 10 GAUGE, (0.134" (3.4 MM) THICK NOMINAL), WITH 1" X 1" X 1/8", (25 MM X 25 MM X 3 MM) ANGLE STIFFENERS ON 4" (101.6 MM) SPACING MINIMUM TO MAINTAIN DUCTING RIGIDITY. ACCESS PORTS MUST BE ADDED TO THE DUCTING AFTER THE INLET FILTER AND APPROXIMATELY 10' (3000 MM) THEREAFTER ON THE HORIZONTAL RUN. ACCESS PORT MUST BE ADDED TO THE DUCT SETTING ON TOP OF THE INLET SILENCER. LEAKS IN DUCTING OR DUCTING INTERCONNECTS CAN NOT BE ALLOWED AND MAY CAUSE ENGINE DAMAGE OR FREQUENT ENGINE MAINTENANCE. CARE SHOULD BE TAKEN IN THE DESIGN TO ENSURE THAT NO LOOSE OBJECTS CAN BE INGESTED INTO THE ENGINE. THE DESIGN PRESSURE DIFFERENTIAL FOR THE DUCTING SHOULD BE NO LESS THAN 20" (508 MM) H₂O. THE DESIGN PARAMETERS EXPLAINED HERE HAVE BEEN FOUND TO BE ADEQUATE FOR ALL OPERATING CONDITIONS OF THE ENGINE. ANY DESIGN MUST CONSIDER CONDITIONS OTHER THAN NOMINAL, OPERATING SUCH AS LIGHT-OUT, FLAME-OUT OR COMPRESSOR SURGE/STALL.

3.8 FOR DETAILED INFORMATION ON THE AIR FILTER, REFER TO VENDOR SUPPLIED INSTALLATION, OPERATION AND MAINTENANCE MANUAL INCLUDED IN THE SOLAR OPERATION AND MAINTENANCE SUPPLEMENTAL DATA. AIR FILTER TECHNICAL DATA:
- INITIAL CLEAR DELTA P: 1.0" (25.4 MM) H₂O
- HIGH DELTA P SWITCH SET POINTS: 5" (127 MM) H₂O ALARM, 7" (178 MM) H₂O SHUTDOWN

REVISION	DESCRIPTION	DATE	APPROVED
NO	INITIAL SUBMITTAL PER PRD 75048	7-30-2004	J.P.M.
A	REVISED TO REFLECT ENGINEERING CHANGES PER PRD 75048	10-28-2004	J.P.M.
B	REVISED TO REFLECT AS-SHIP-TO CONFIGURATION PER PRD 75048	12-17-2004	J.P.M.

SANTEE COOPER DWG#
G-R1-TG-105-00-0001-001

CUSTOMER
SANTEE COOPER
RICHLAND COUNTY LANDFILL
GAS-TO-ENERGY FACILITY

☒	SUBMITTED FOR REVIEW	DATE 7-30-04
☐	CERTIFIED AS NOTED WITH	DATE
☒	CERTIFIED FOR CONSTRUCTION	DATE 12-17-04

SYMBOLS

③	DENOTES EXTERNAL CONNECTION POINT
②	DENOTES REVISION
①	DENOTES COMPONENT OR INSTRUMENTATION
④	DENOTES DRAWING ZONE LOCATION

FIG. NO.
66195

THIS COPYRIGHTED WORK AND THE INFORMATION HEREIN IS THE PROPERTY OF SOLAR TURBINES INC. SOLAR TURBINES INC. IS NOT RESPONSIBLE FOR ANY REPRODUCTION, IN ANY FORM, WITHOUT EXPRESS, WRITTEN PERMISSION FROM SOLAR TURBINES INC. OR ITS AFFILIATES. NO USE EXCEPT THAT FOR WHICH IT IS LOANED, IS PERMITTED.

APPROVED

PROJECT MGR. DATE 7-30-04

ASSISTANT MGR. DATE 7-30-04

ENGINEER DATE 7-30-04

NA

PROJECT ENG. DATE 7-30-04

NA

HYDRO-MECH. ENG. DATE 7-30-04

NA

QA DESIGNER DATE 7-23-04

J.P. MUYCO 7-23-04

SOLAR PROJECT IDENT. NO. 2-75681

CUSTOMER IDENT. NO. SEE ABOVE

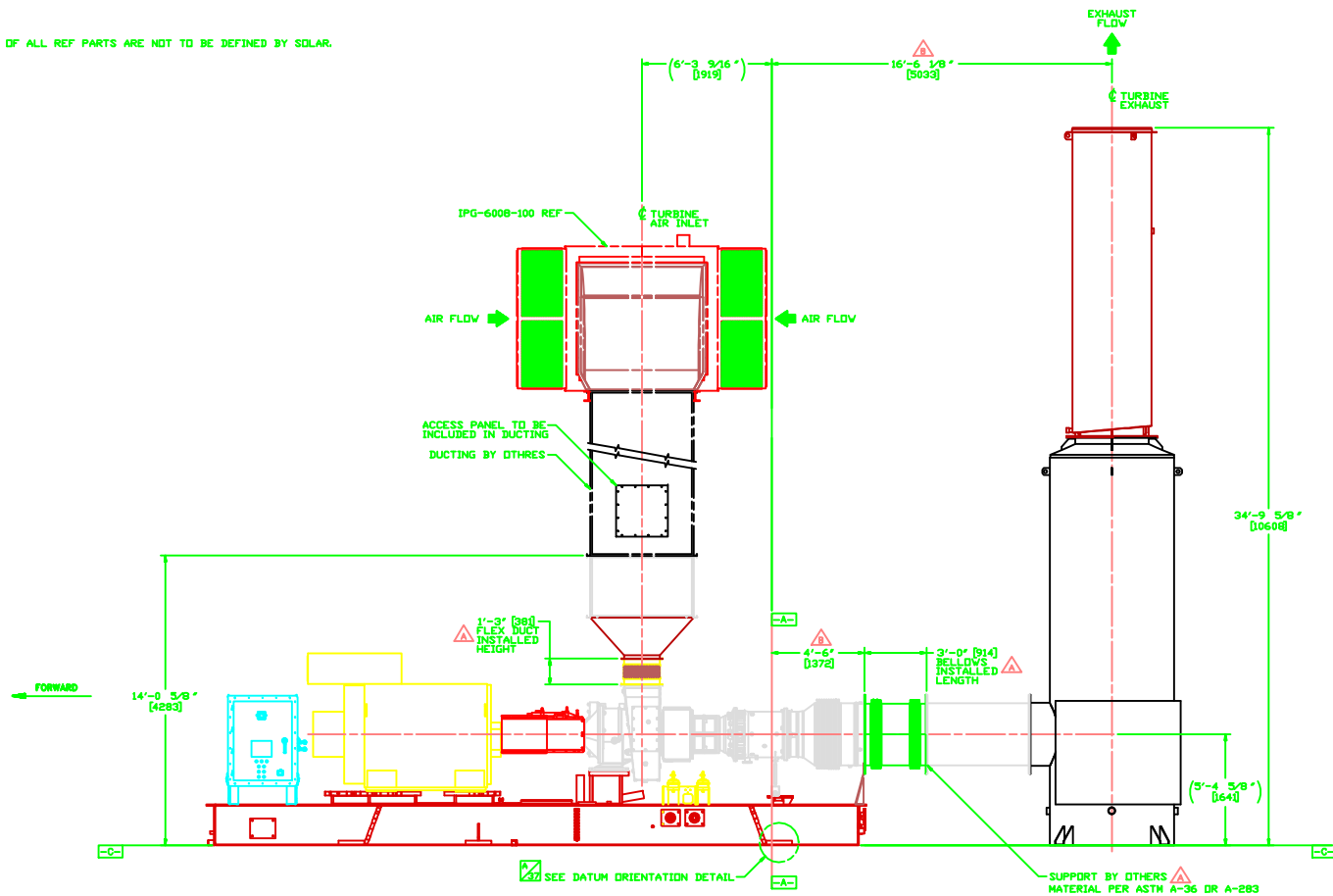
DRAWING TITLE
TAURUS 80
POWER GENERATION
AUXILIARY EQUIPMENT
MECHANICAL INSTALLATION

Solar Turbines
A Delmar Group

DRAWING NO. 75681-149222

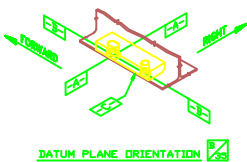
REV B
SHEET 1 OF 8

NOTE: POSITION OF ALL REF PARTS ARE NOT TO BE DEFINED BY SOLAR.



LEFT SIDE ELEVATION

SANTEE COOPER DWG#
G-R1-TG-105-00-0001-003



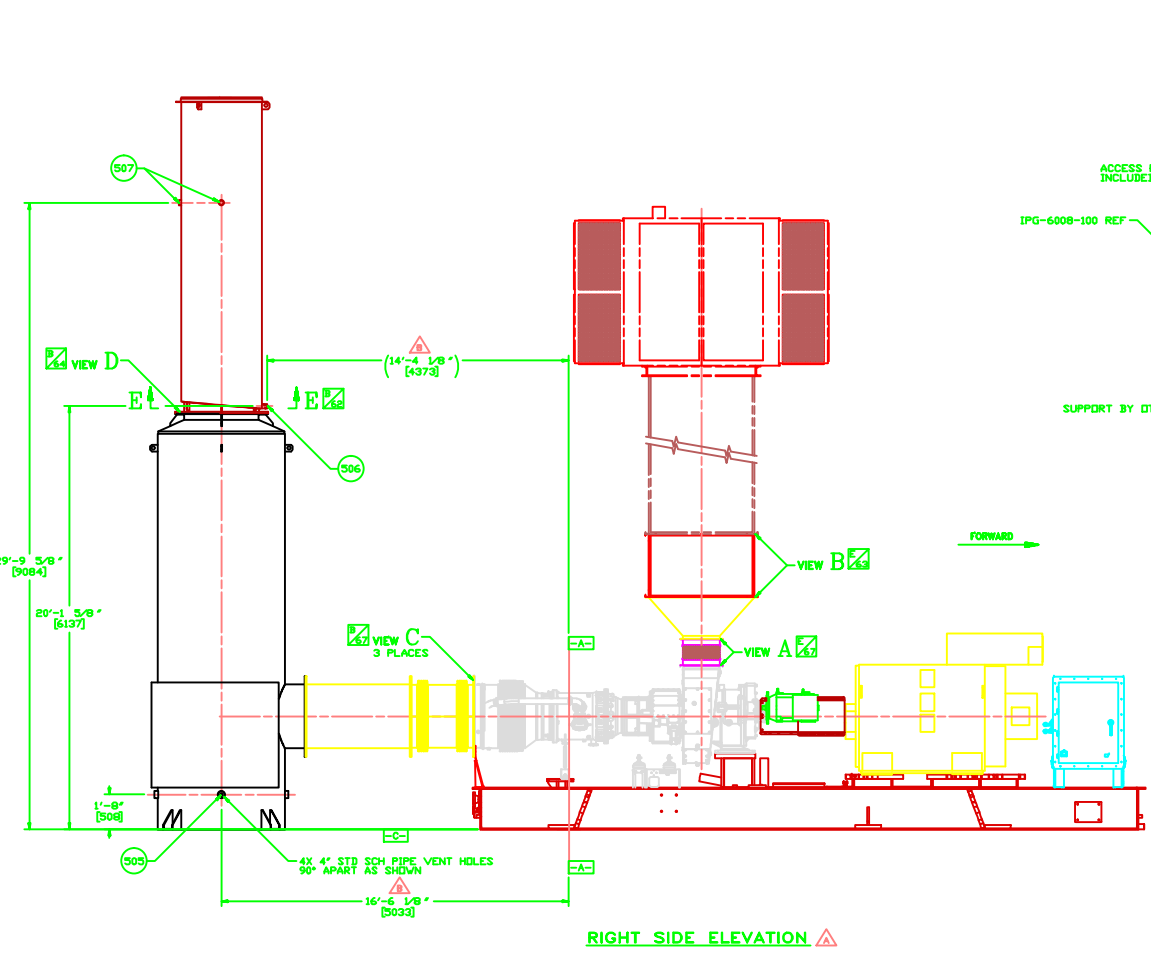
- A- PACKAGE TIE-DOWN SLOT
- B- PACKAGE TIE-DOWN SLOT
- C- BOTTOM OF SKID

NOTE: ALL DATUMS ARE FROM THE BOTTOM OF THE SKID.

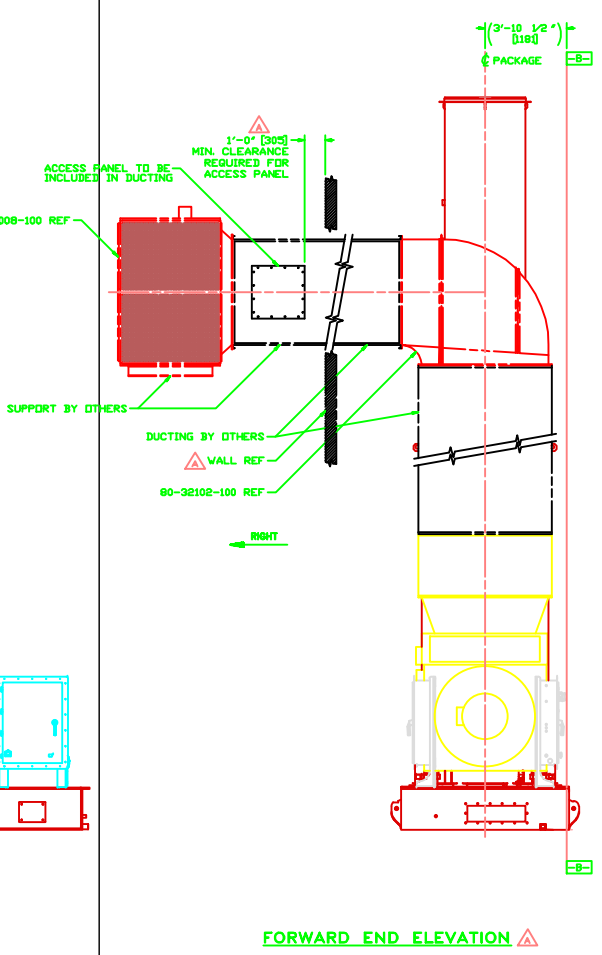
Solar Turbines
A Solar Turbines Company

75681-149222 REV B
SHEET 3

TABLE 1 - EXTERNAL CONNECTIONS			
ITEM	DESCRIPTION	PREFERENCE NOTE	ZONE
500	EXHAUST SILENCER DRAIN 1" NPT (CAPPED), SCH 40 CARBON STEEL		B/48
504	EXHAUST RAIN STACK DRAIN 2" NPT (CAPPED), SCH 40 CARBON STEEL		D/47
505	EXHAUST EMISSIONS TEST PORTS, HALF COUPLINGS Q2 PLACES 2" NPT		F/48



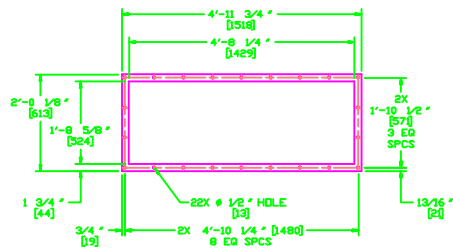
RIGHT SIDE ELEVATION



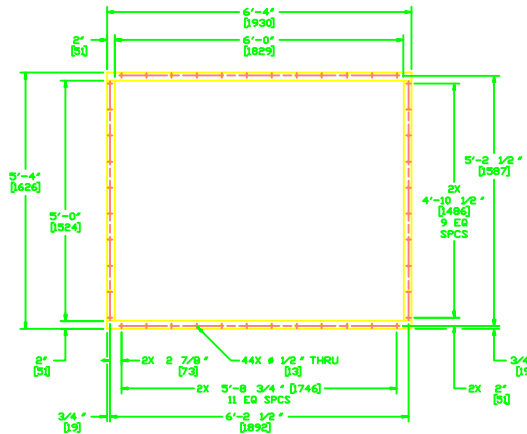
FORWARD END ELEVATION

SANTEE COOPER DWG#
G-R1-TG-105-00-0001-004

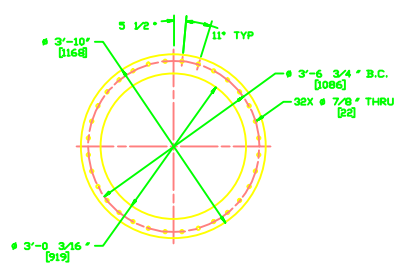
Solar Turbines
A Solar Turbine Company
75681-149222
REV B
SHEET 4



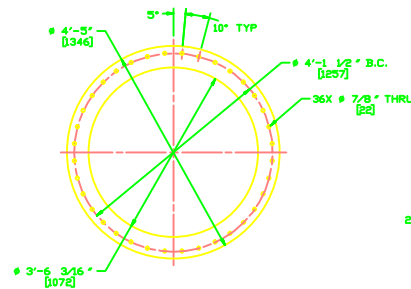
VIEW A



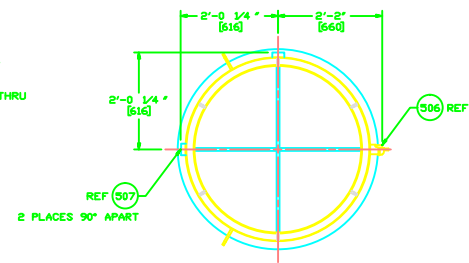
VIEW B



VIEW C



VIEW D

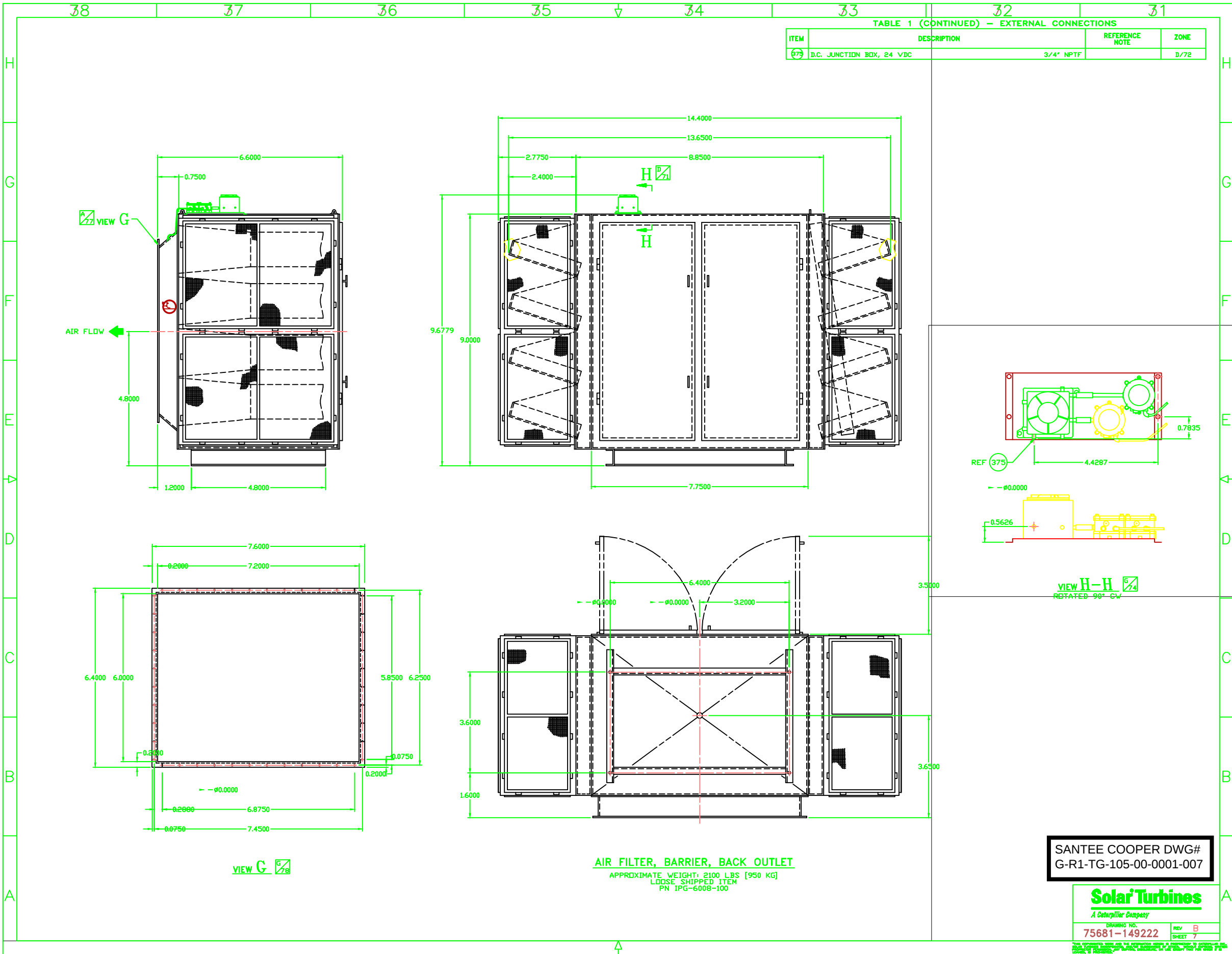


VIEW E-E

SANTEE COOPER DWG#
G-R1-TG-105-00-0001-006

Solar Turbines
A Solar Turbines Company

75681-149222 REV B
SHEET 8

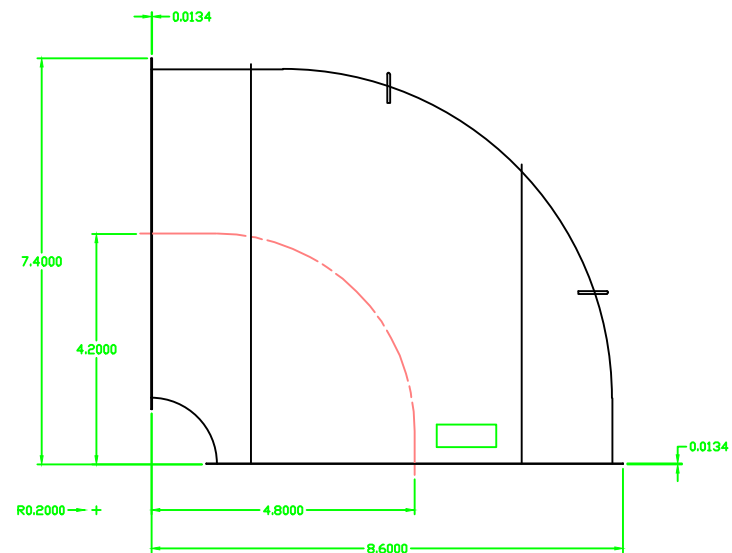


SANTEE COOPER DWG#
G-R1-TG-105-00-0001-007

Solar Turbines
A Caterpillar Company

DRAWING NO.
75681-149222

REV
B
SHEET
7



SANTEE COOPER DWG#
G-R1-TG-105-00-0001-008

Solar Turbines

A Caterpillar Company

DRAWING NO. 75681-149222	REV B
	SHEET 8

*THIS DOCUMENTED WORK AND THE INFORMATION HEREIN IS PROPRIETARY TO CATERPILLAR INC. ALL RIGHTS RESERVED. NO PART OF THIS DOCUMENT OR THE INFORMATION CONTAINED HEREIN MAY BE REPRODUCED OR TRANSMITTED IN ANY FORM OR BY ANY MEANS, WITHOUT PERMISSION IN WRITING FROM CATERPILLAR INC. ANY REPRODUCTION OR TRANSMISSION OF THIS DOCUMENT OR THE INFORMATION CONTAINED HEREIN WITHOUT PERMISSION IS STRICTLY PROHIBITED. ANY COPIING, REPRODUCTION, OR USE WITHOUT THE PERMISSION OF CATERPILLAR INC. IS PROHIBITED.